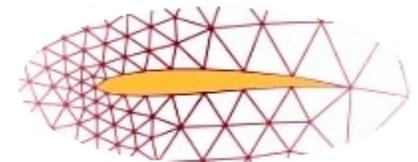
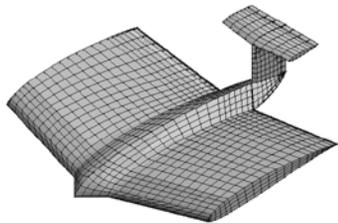


On the Effect of Winglets on the Performance of Micro-Aerial-Vehicles

Jarmo T. Mönttinen, Helen L. Reed,
Kyle D. Squires, and William S. Saric
**MECHANICAL AND AEROSPACE
ENGINEERING**

The 1st Annual REAS Conference
Arizona State University
September 4, 2003

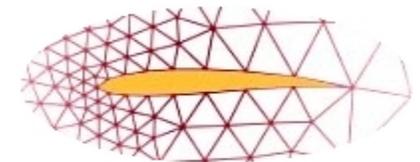
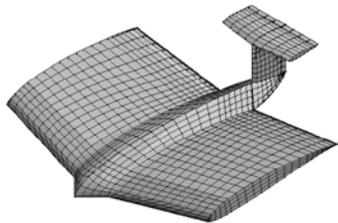


Winglets in everyday use

- Boeing business jet

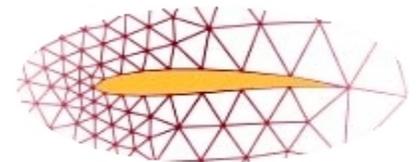
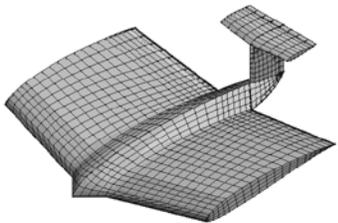


- Air Berlin 737-800



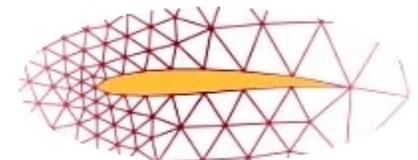
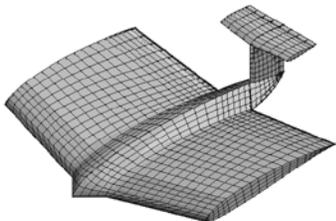
Outline

- **Definition of an MAV**
- **Aerodynamic considerations**
- **Winglets**
- **Computational approach**
- **Current results**
- **Ongoing work**
- **Conclusions**



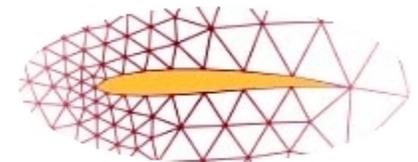
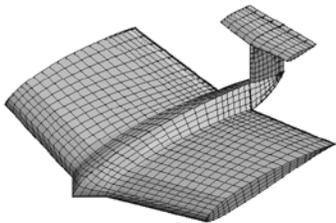
What is an MAV?

- **Maximum dimension: 150 mm**
- **Mass 60-80 grams**
- **Operates at $Re = 50,000 - 200,000$**
- **Capable of carrying a payload (20g)**
- **Equipped with high tech surveillance equipment**
- **Multi-disciplinary research problem**
 - » **Airframe**
 - » **Propulsion**
 - » **Avionics**



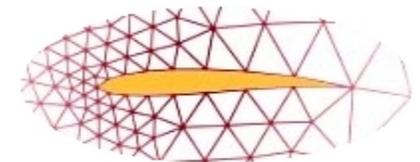
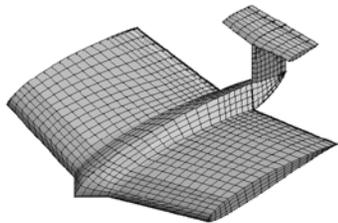
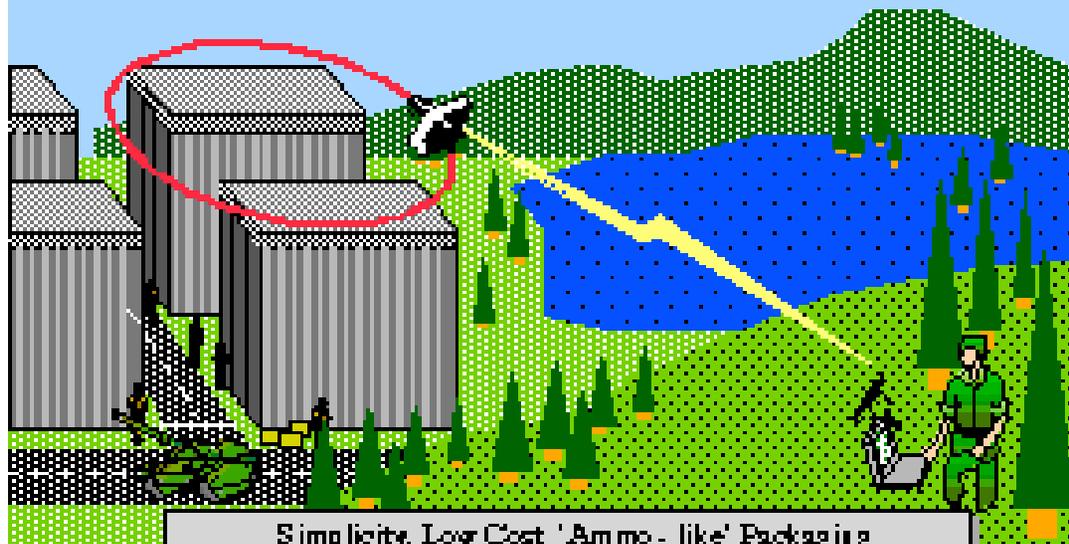
MAV missions

- **Surveillance**
- **Sensor placement**
- **Detection of chemical and biological threats**



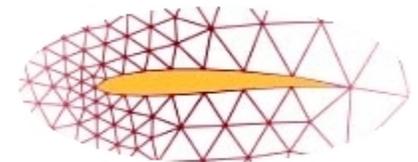
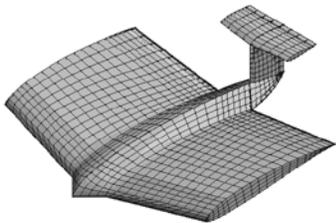
MAV operation

- Range 10 km
- Speed: 10-20 m/s
- Duration: 20-60 min
- Day-Night Imaging (near real time)



Problem Characteristics

- **Low aspect ratio and short span**
 - Size restriction
- **Low Reynolds number**
 - Size restriction and low speed
- **Unsteady freestream**
 - Low altitude

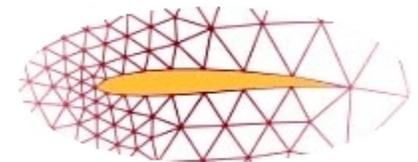
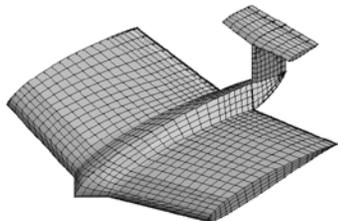


Aerodynamics

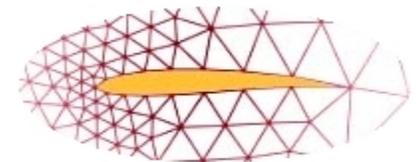
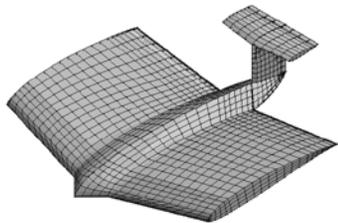
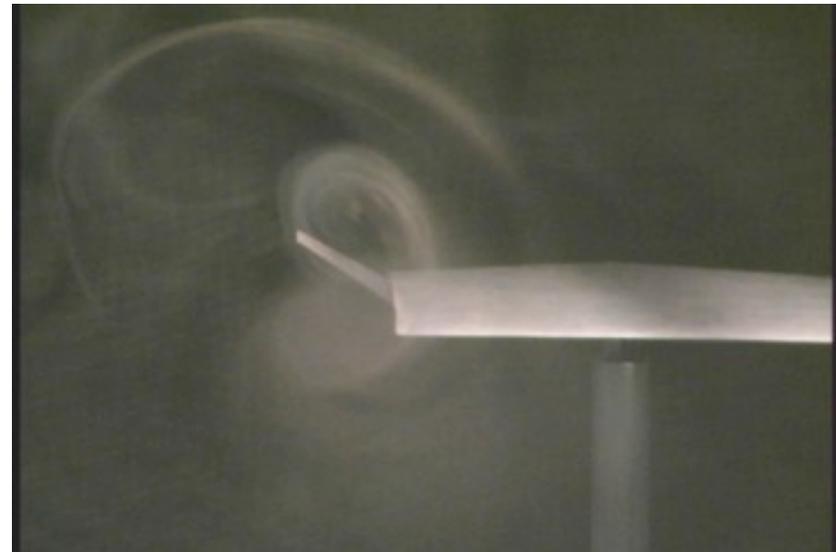
- **Maximize lift**
 - Payload
 - Range
 - Endurance
- **Induced drag**
 - Tip-vortices



Dryden Flight Research Center ECN 3831 Photographed 1974
B-727 vortex study NASA photo

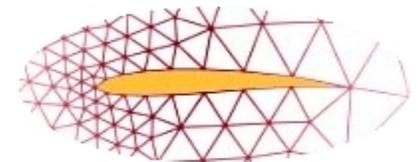
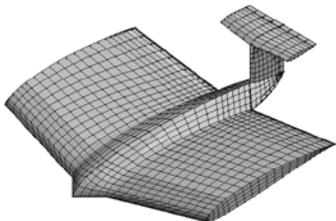


Why winglets?



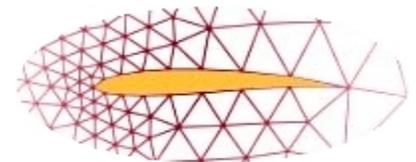
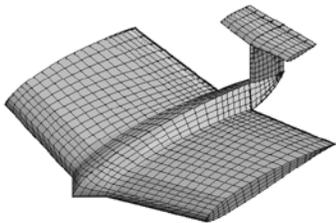
Effect of Winglets

- **Limited downwash**
 - Increase in lift
 - » Important to enable payload carriage
 - Drag reduction
 - » Induced drag, potentially total drag
 - » **Success stories**
 - Sailplanes
 - Transonic aircraft
 - Business jets



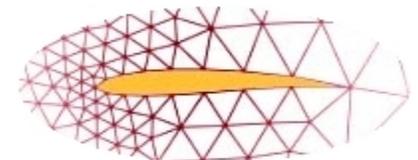
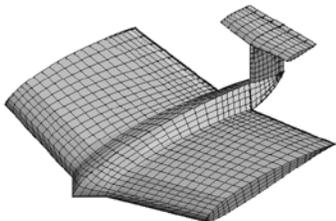
Motivation for the current study

- **Lack of published data for the full Re-range where MAVs operate**
 - Data exists for $Re > 100,000$
- **No computational studies of winglets for MAVs available**
 - Investigate the effect of winglets only
 - Success with other applications
- **Provide a database that can be used by future researchers or MAV-designers**



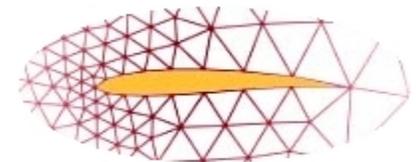
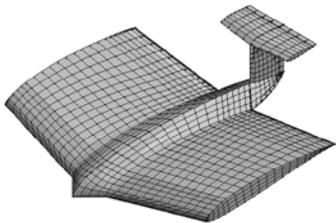
Computational Approach

- **COBALT**
 - Compressible Navier-Stokes solver
 - » Run incompressible at Mach = 0.1
 - Finite Volume Method
 - Riemann Solver
- **Gridgen**
 - Block-based grid generator

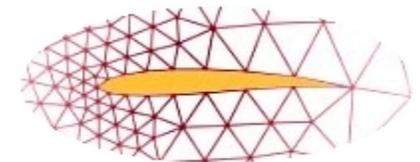
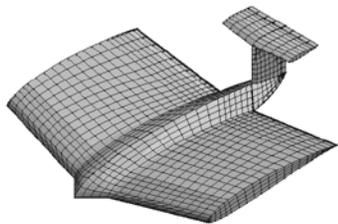
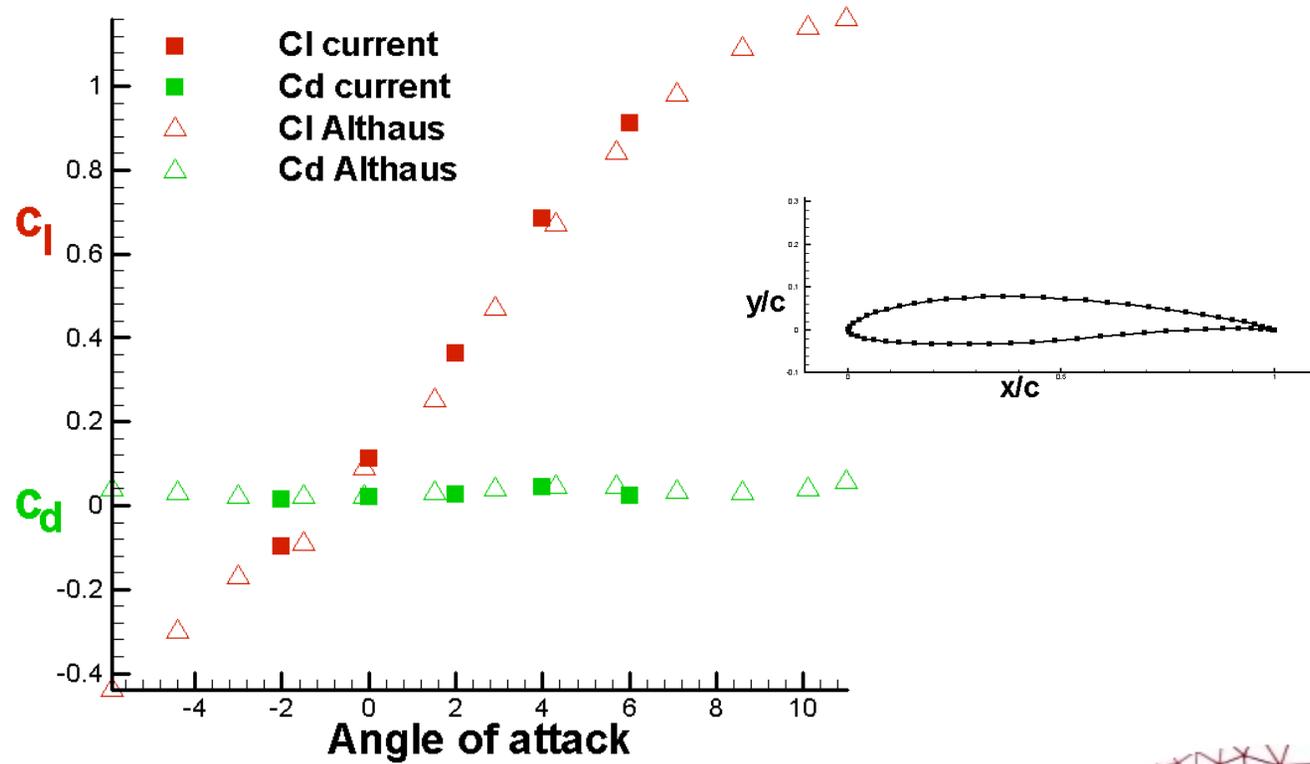


Validation and Verification

- **Required** for reliable computations!
- **Validation**
 - Does the computation model real life?
 - Compare with experimental results
- **Verification**
 - Are the equations solved correctly?
 - Compare with analytical solutions or other computational results

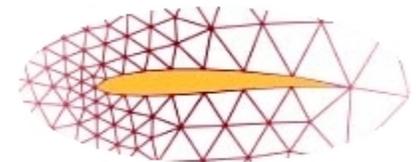
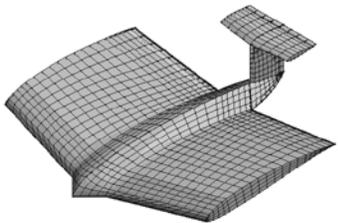


Eppler 211, Re= 60,000 Validation



3-D work

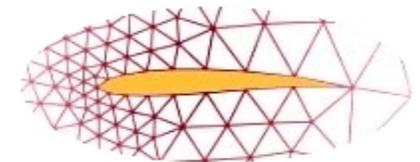
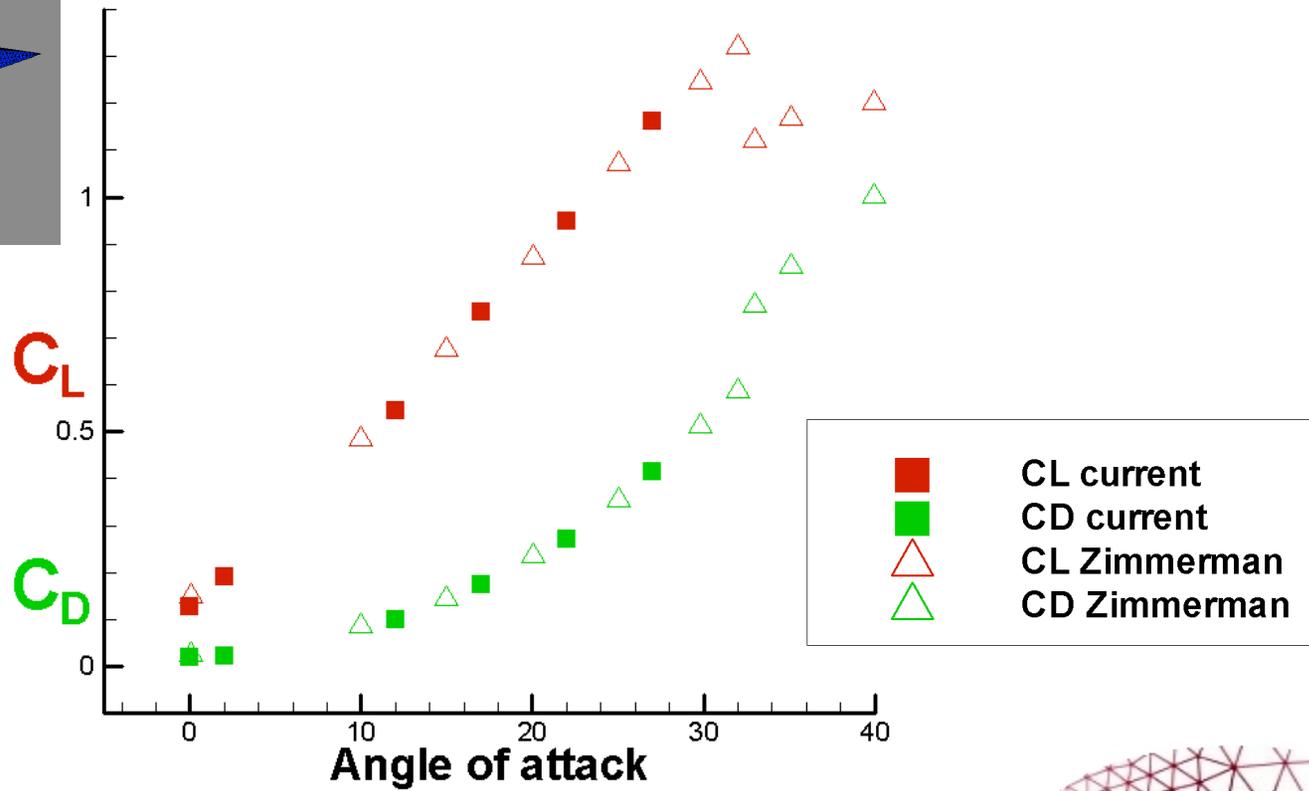
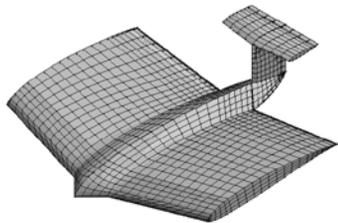
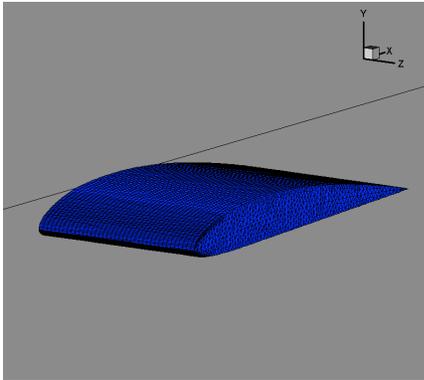
- **Baseline model vs. wing-winglet model**
- **How do the winglets affect**
 - Lift
 - Drag
 - Lift-to-drag-ratio
 - Stability and controllability



Clark-Y wing

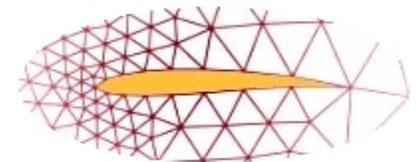
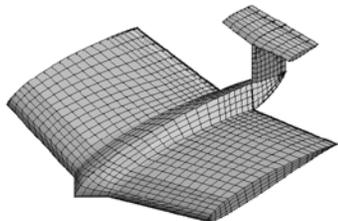
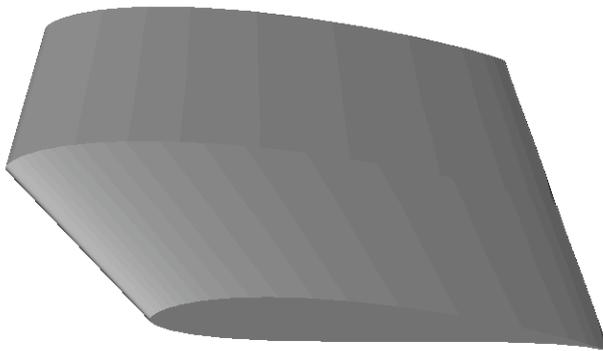
AR = 1, Re = 860,000

Validation



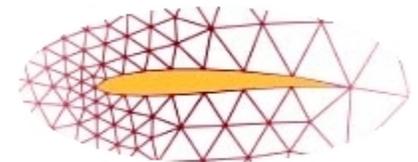
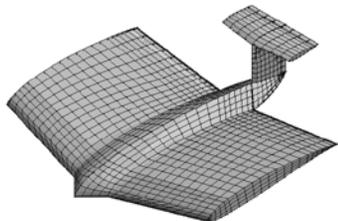
Baseline Model

- Eppler 212
- Root chord 150 mm
- Tip chord 127 mm
- Straight trailing edge
- Span 150 mm



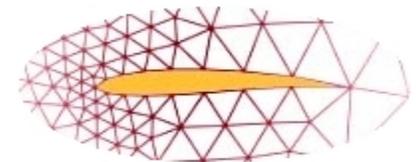
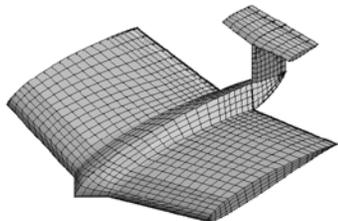
Winglet 4b

- 25.4 mm span
- 78 mm tip chord
- Leading edge sweep angle 26.6°
- 90° cant angle
- Thickness 1.6 mm

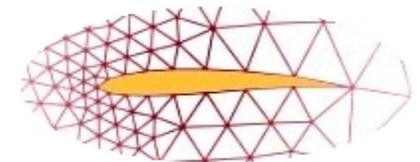
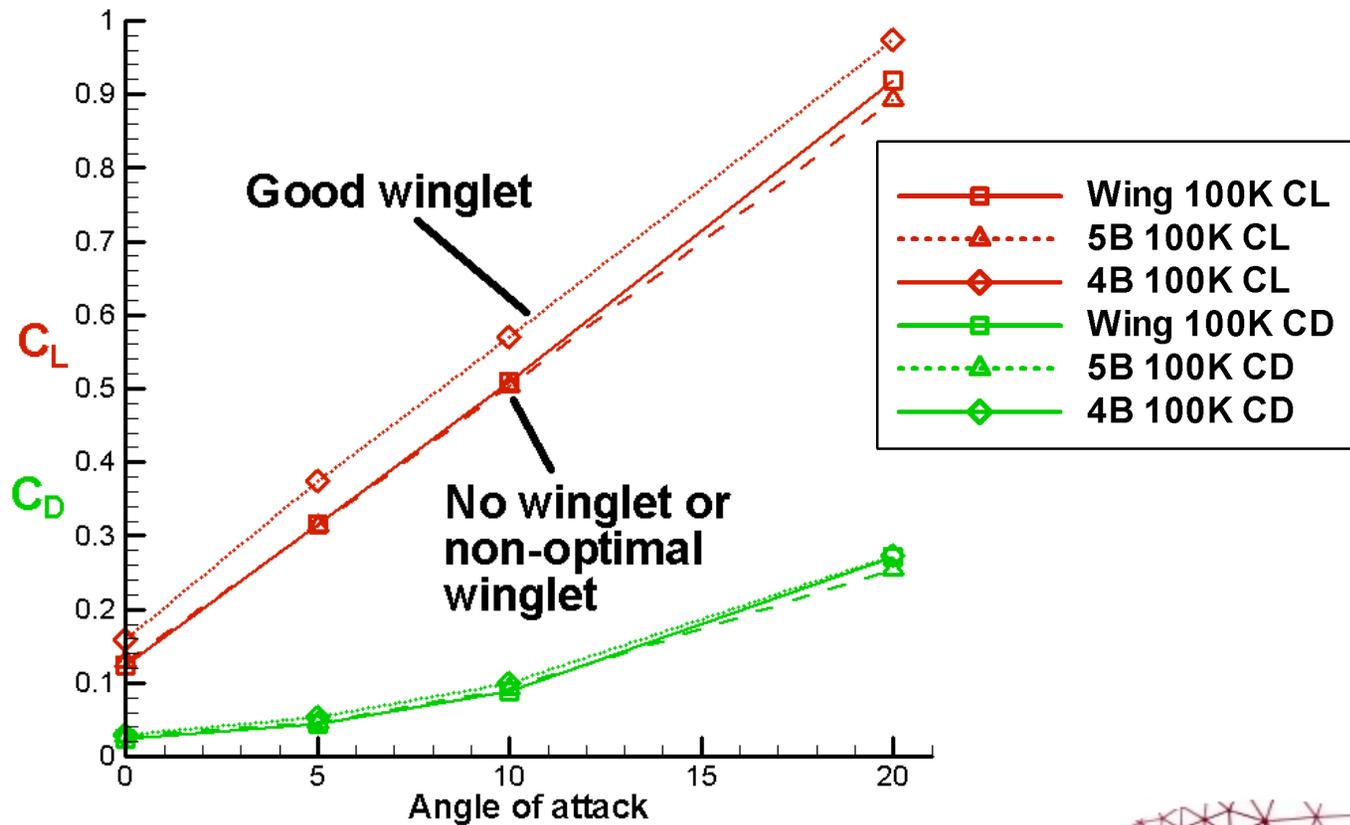
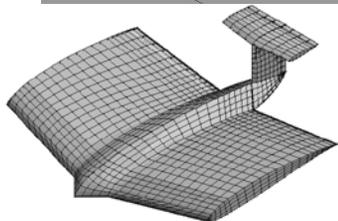
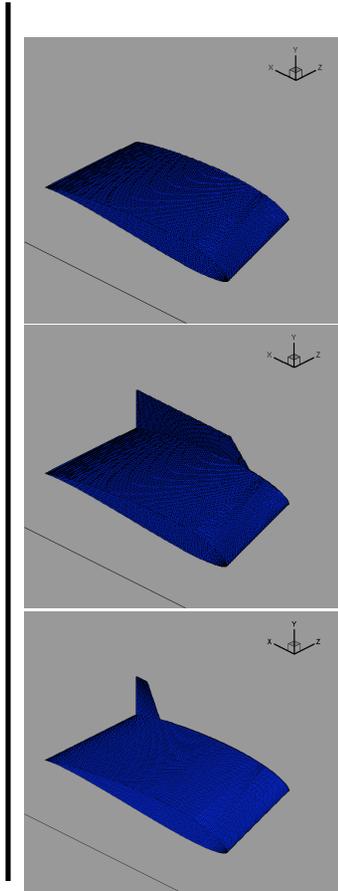


Winglet 5b

- 25.4 mm span
- 8 mm tip chord
- Leading edge sweep angle 26.6°
- 90° cant angle
- Thickness 1.6 mm

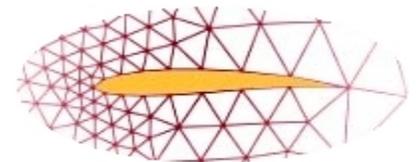
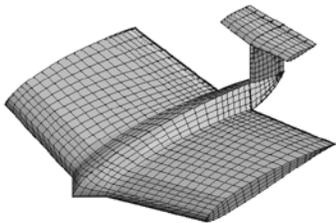


Effect of Winglets, $Re = 100,000$



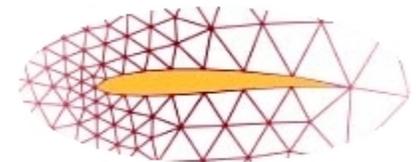
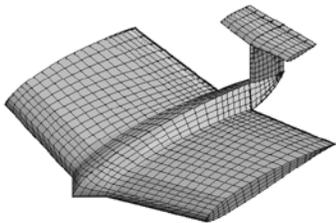
Ongoing work

- **Complete computations to optimize winglets**
 - $Re = 50,000-200,000$
- **Analyze the effect of winglets on the stability of an MAV**
- **Provide guidelines for choosing winglets for MAV-applications**



Conclusions

- **MAV definition**
 - Aerodynamic considerations
- **Computational method**
 - Validation and verification
- **Winglets can improve both the lift and the lift-to-drag-ratio when properly chosen**
 - Careful: possibility of adverse effects!



Acknowledgements

- Dr. Kyle D. Squires
- AFOSR
- ASU Wind Tunnel Complex

