

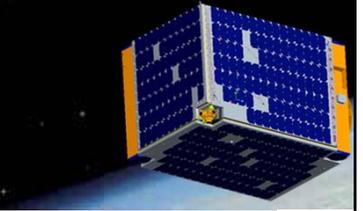
# An Iterative Subsystem-Generated Approach to Populating a Satellite Constellation Tradespace

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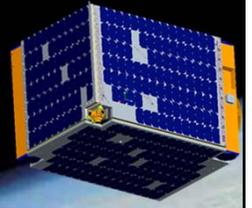


# Outline



- Introduction to Tradespace Exploration
- Defining the Tradespace
- Methods:
  - Populating the Tradespace by Iteration
  - Utility Attributes
  - Design Variables and Cost Drivers
- Exploring the Tradespace by Attribute
- Multi Attribute Tradespace Exploration (MATE)
- Conclusions

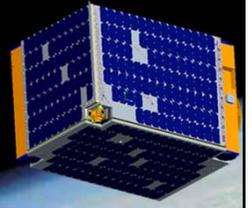
# Tradespace Exploration



## Objectives:

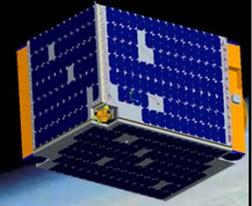
1. Compare potential designs on a common basis (and gain insight in the process)
2. Identify high value (utility/cost) designs to be examined in further detail
3. Examine the performance/cost/schedule implications of system requirements (which are easily achieved and which are significant cost drivers?)

# Tradespace Exploration

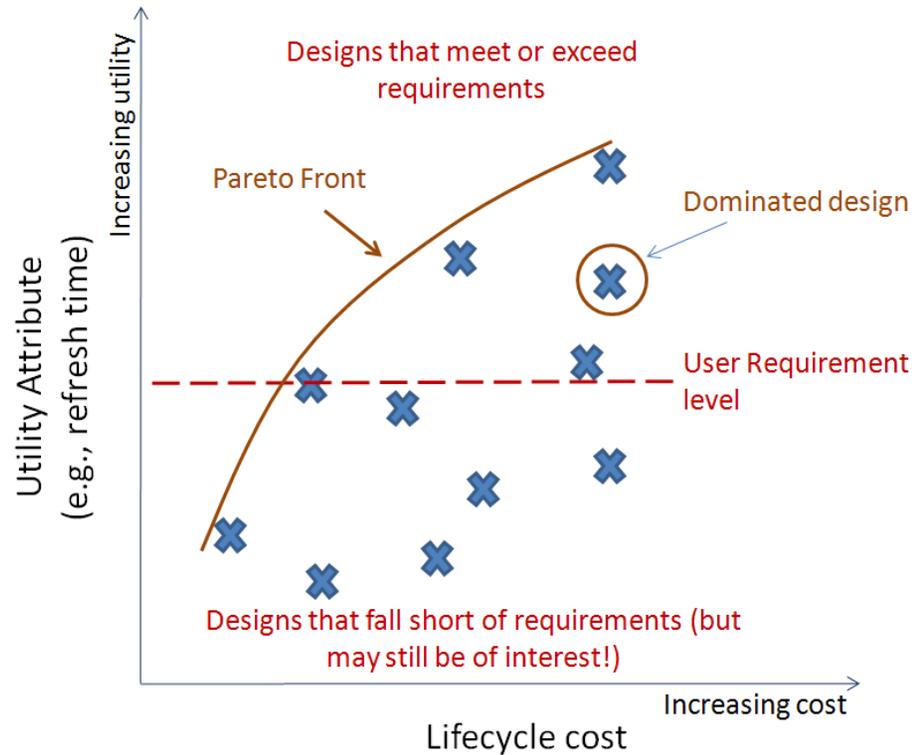


- Cost vs. performance tradeoff:
  - Which is better – a system that meets most objectives at a reasonable cost, or a system that meets all requirements at double the cost?
- Process rather than point analysis: insight gained through analysis is more important than result.
- Complex space missions often have many competing design alternatives, each comprising trade-offs across multiple subsystems.
  - It is not straightforward to span the tradespace by simply selecting design independent variables at either the system or subsystem level.
- How do we deal with these interactions while efficiently mobilizing domain expert knowledge?

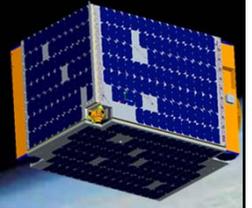
# Example Tradespace



- Increasing *value* is composed of:
  - increased utility (increasing along the y-axis, and
  - lower cost (decreasing along the x-axis).
- Pareto Front represents designs that have:
  - the highest utility for a given cost, or
  - the lowest cost for a given utility.
- Designs falling closer to the Pareto Front are higher in value.
- Designs falling farther below the Pareto Front are dominated by higher value designs.

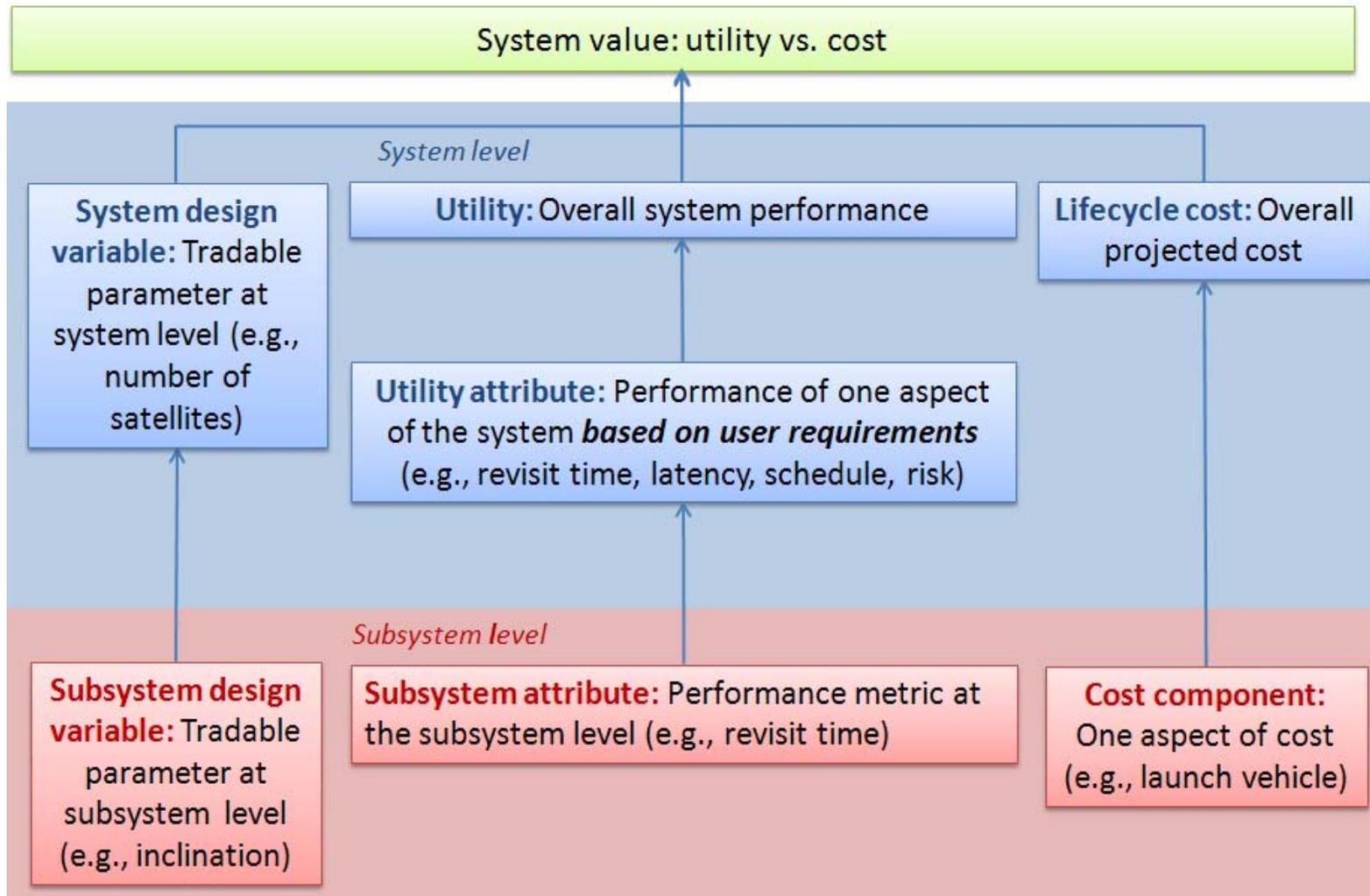
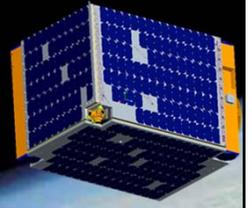


# The Mission



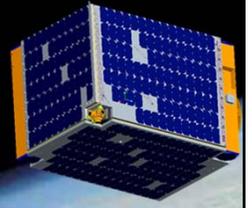
- Earth observation constellation with global coverage and requirements for:
  - high observation times
  - frequent revisits
  - fast downlinks
  - with high reliability for multiple points on Earth
- How these attributes should be traded off versus each other or versus costs?
- How should peak performance be valued versus mean, median, or worst case performance?
- How should global coverage be valued versus coverage over a particular area of interest?
- How much value is derived from additional ground stations as compared with additional satellites?

# Terminology & Hierarchy



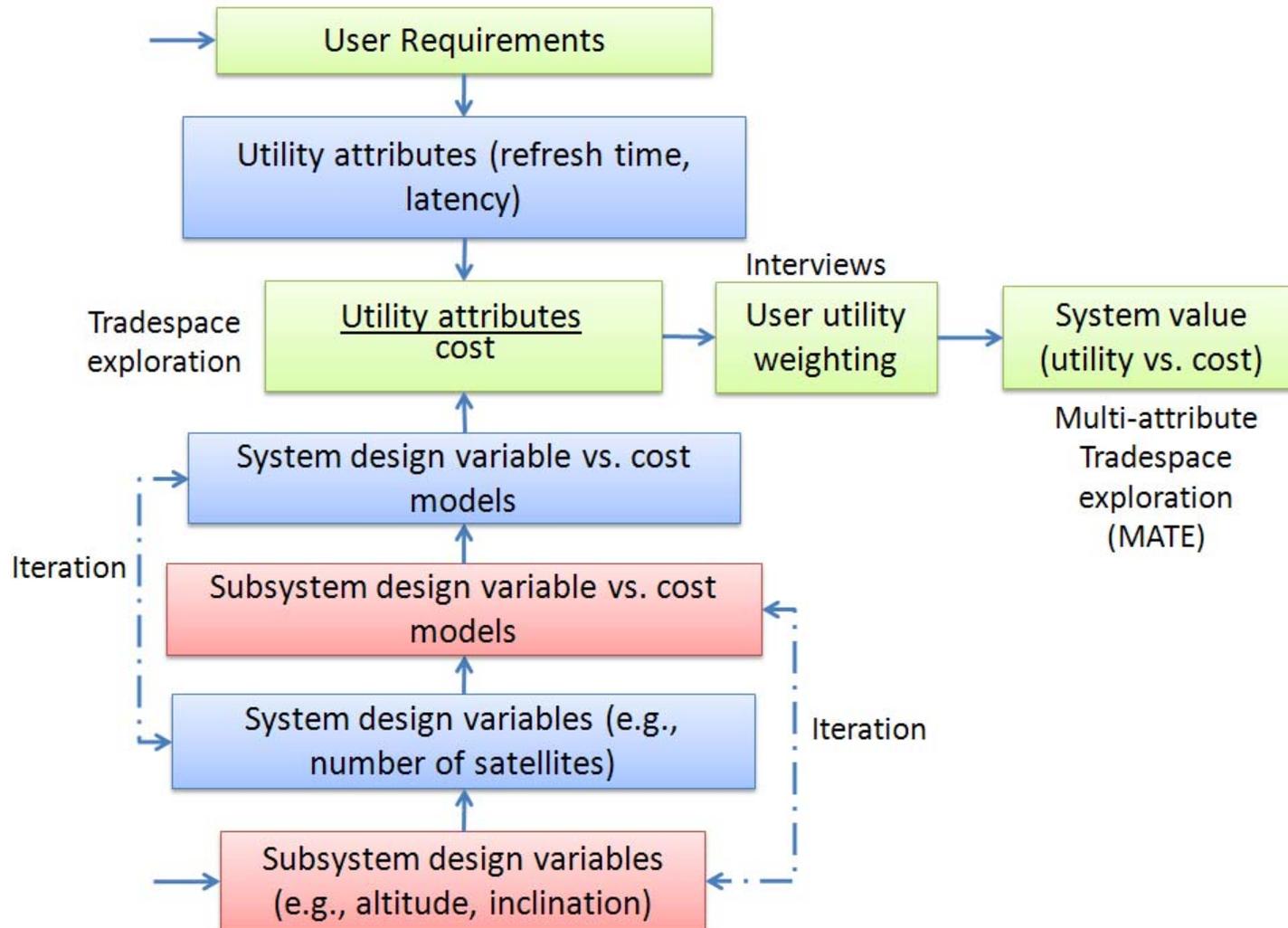
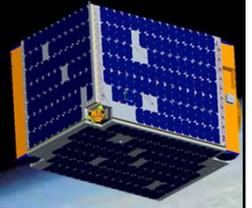
Methods

# Method



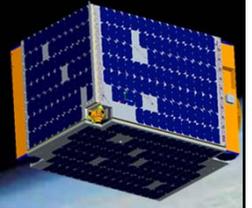
- Subsystem level optimizations performed in parallel by domain experts using their own tools.
- Interactions between subsystems assessed and propagated at system level using key nodal checkpoints.
- Allows independent partitioning of the design space by expertise (e.g., orbit, ground, launch, payload, etc.).
- Assembled system level tradespace automatically captures individual subsystem level trades.
- Tradespace considered multiple architectures, including nanosatellites with low duty cycles and strict power limitations, microsatellites, and hosted payloads on larger satellites.

# Tradespace Exploration Flow



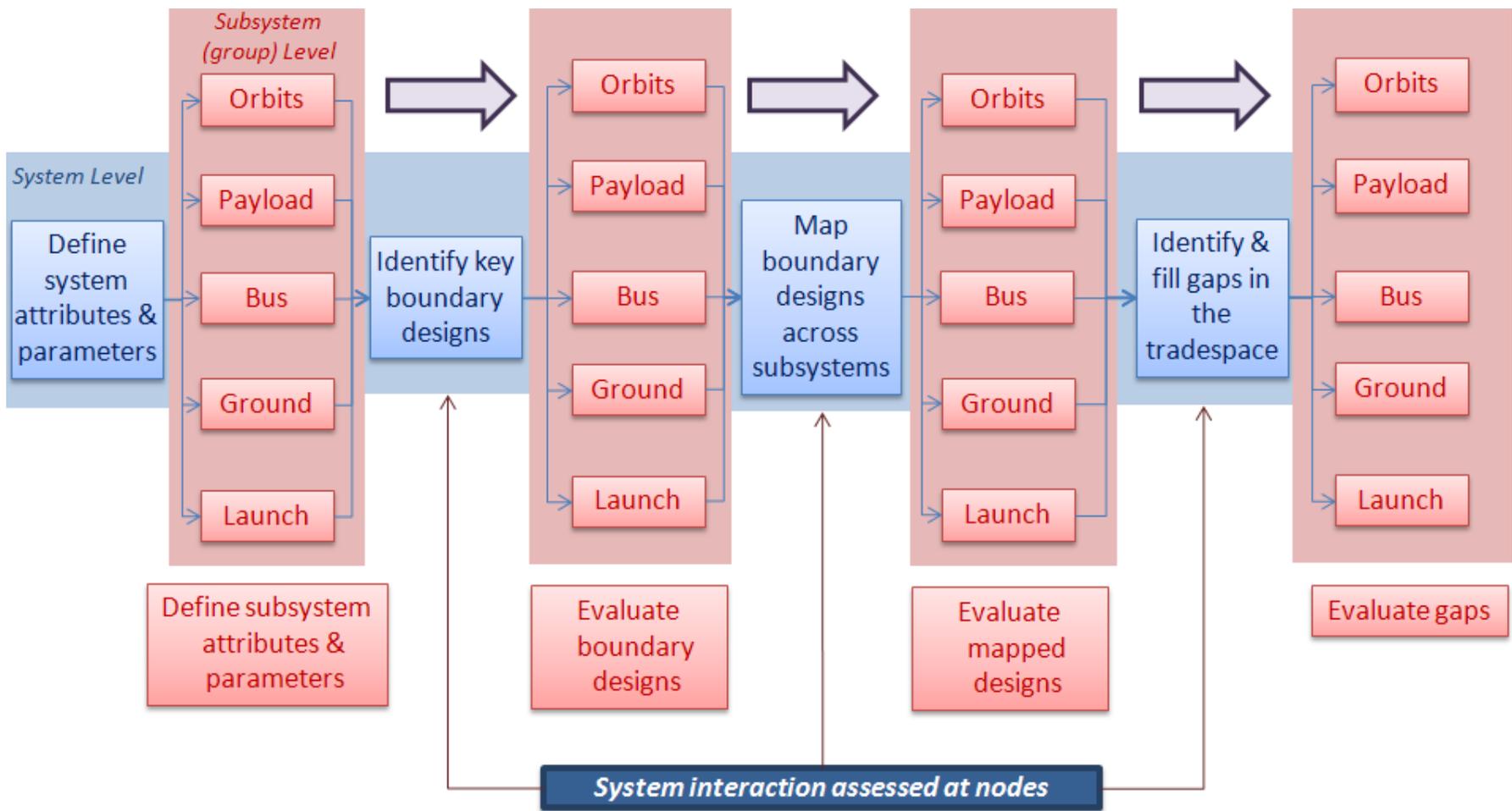
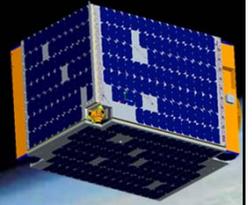
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# Subsystem Level Optimizations



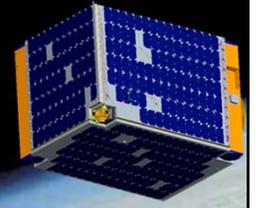
- Subsystem design variables from system design variables e.g., orbit = planes, spacing, inclinations, eccentricities, # per plane, etc.
- Subsystem domain experts are free to use own tools and conduct own optimizations.
- Minimal communication between expert groups is required to ensure that designs are feasible (though not necessarily efficient) across subsystems .
- No need to account for interactions at the subsystem level.
- Note that two subsystems may not produce the same result even with the same goal.

# Populating the Tradespace by Iteration



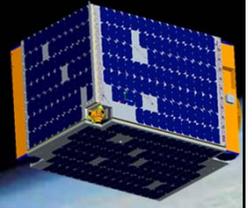
Methods

# Propagating the Tradespace at the System Level

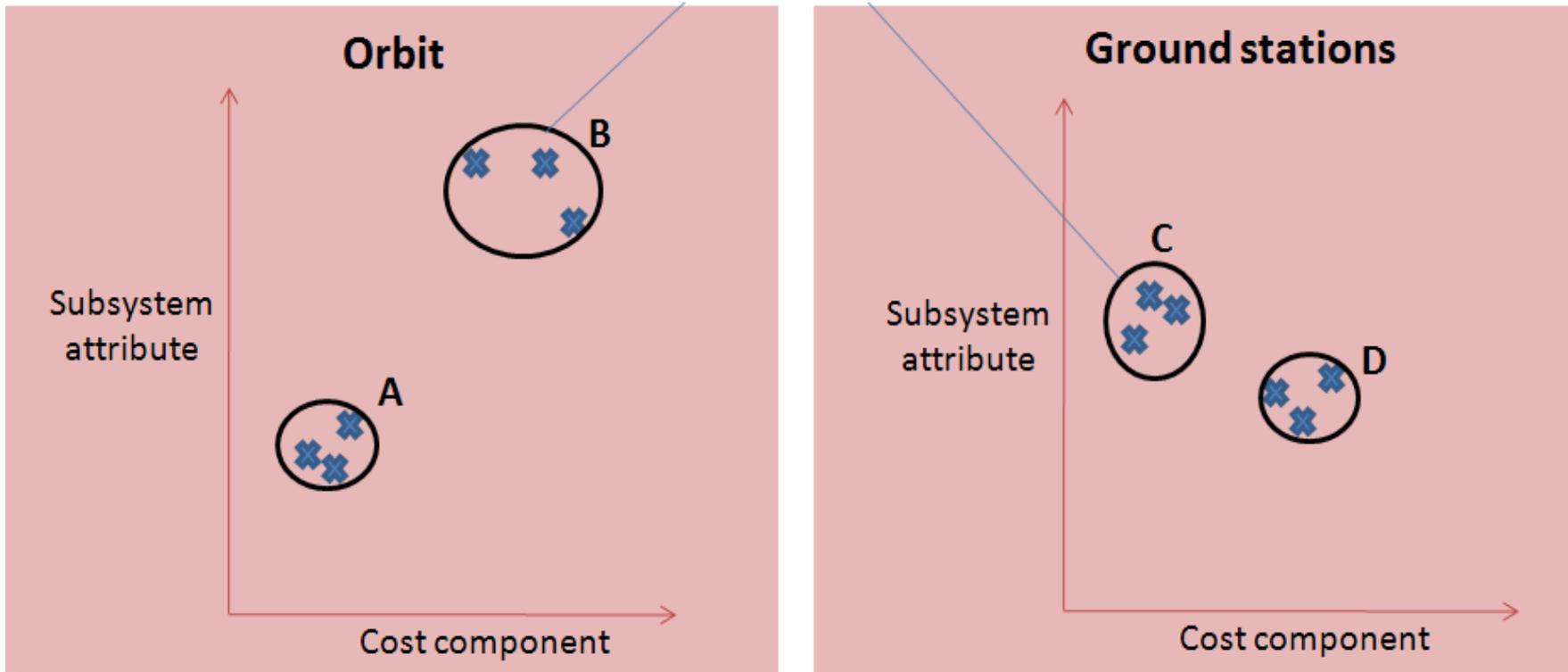


1. Evaluate interactions at system level: i.e., how do subsystem design variables influence system level utilities and costs?
2. Map subsystem-generated design variables across subsystems (possible full factorial expansion, e.g., 5 orbit x 3 ground station x 2 launch =  $5 \times 3 \times 2 = 30$ , but overlaps are likely).
3. Potential for pruning the initial Tradespace.
4. Add missing “gap” designs based on exploration of initial Tradespace – especially when two or more subsystems share design variable elements (e.g., launch analysis suggests cost savings from implementing an orbital configuration).

# Capture Boundary Designs

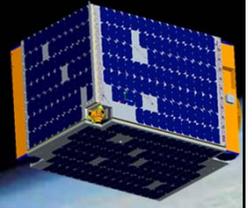


Capture boundary designs  
at subsystem level

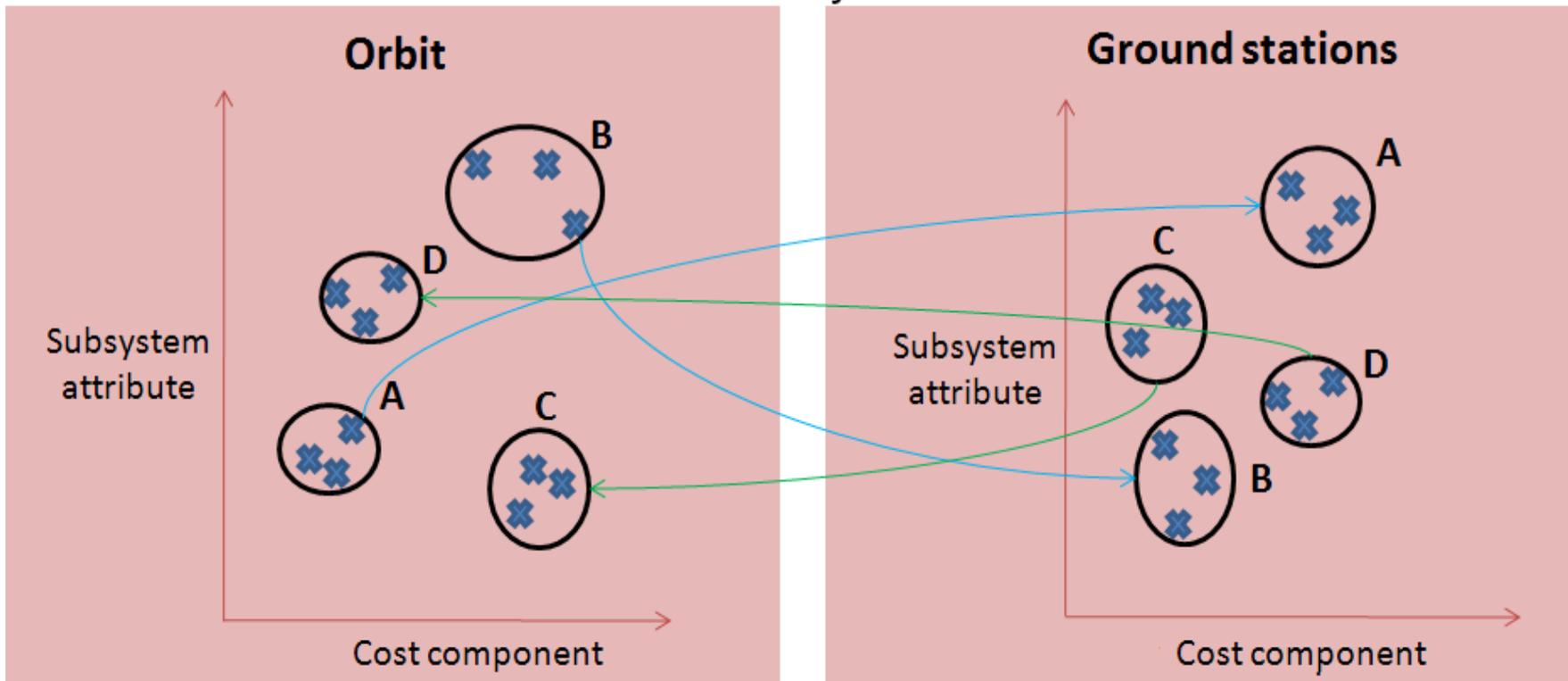


Methods

# Map Across Subsystems

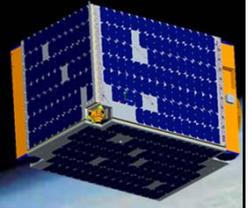


Map boundary designs  
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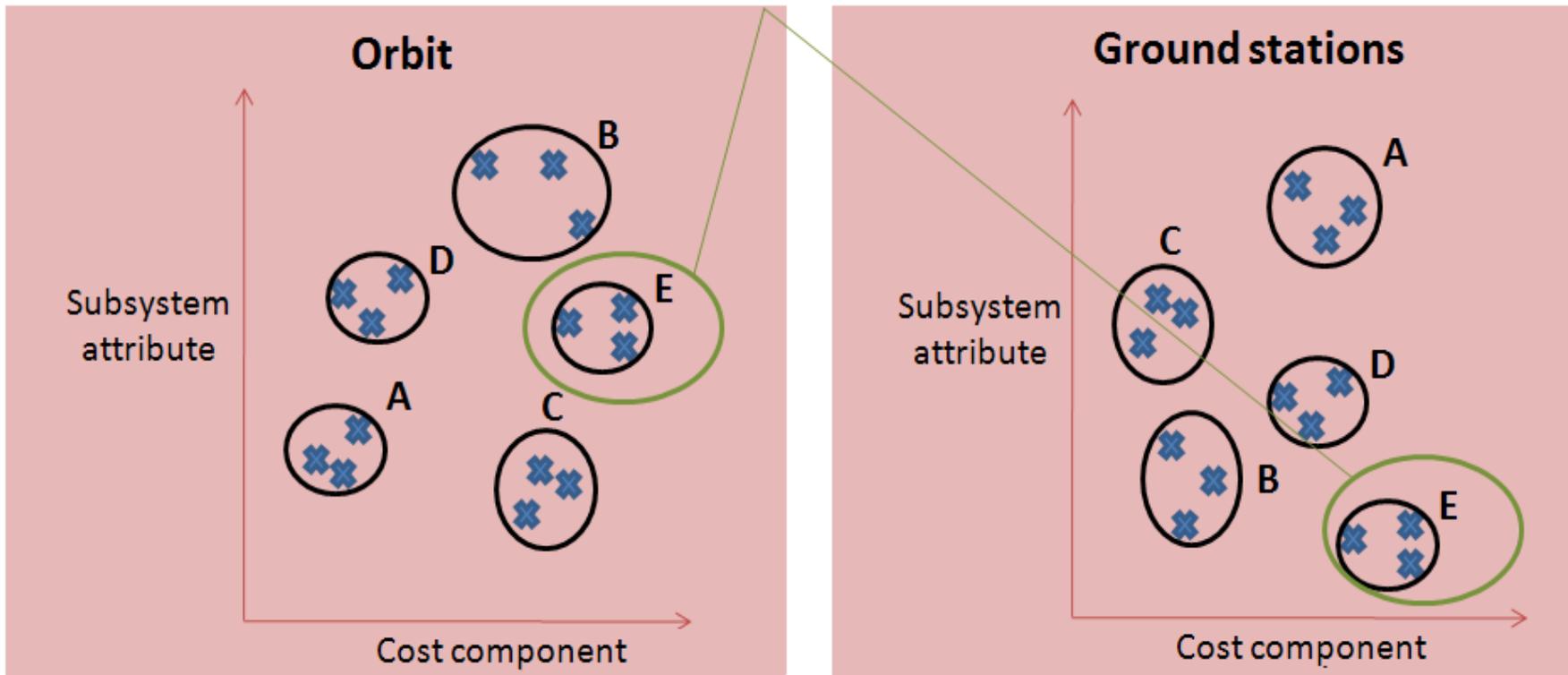


Methods

# Fill Gaps in Tradespace

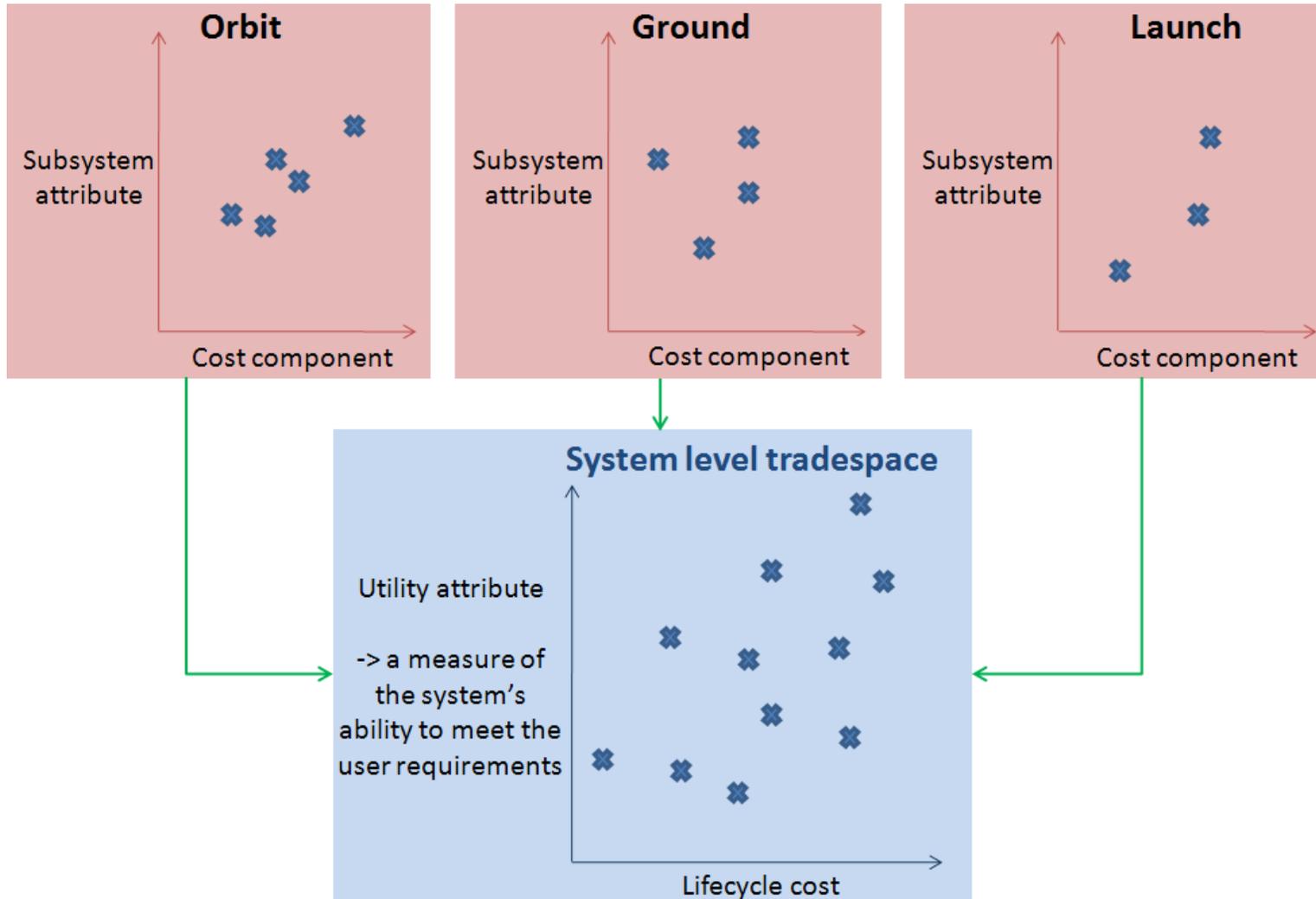
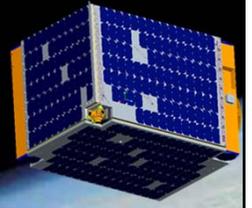


Fill gaps in tradespace: Design E not captured as a boundary design



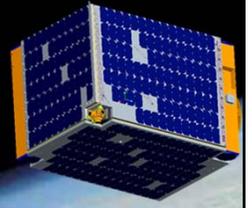
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# Build System Level Tradespace



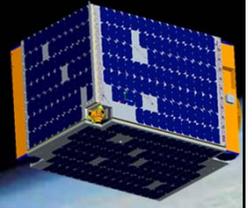
Methods

# Utility Attributes



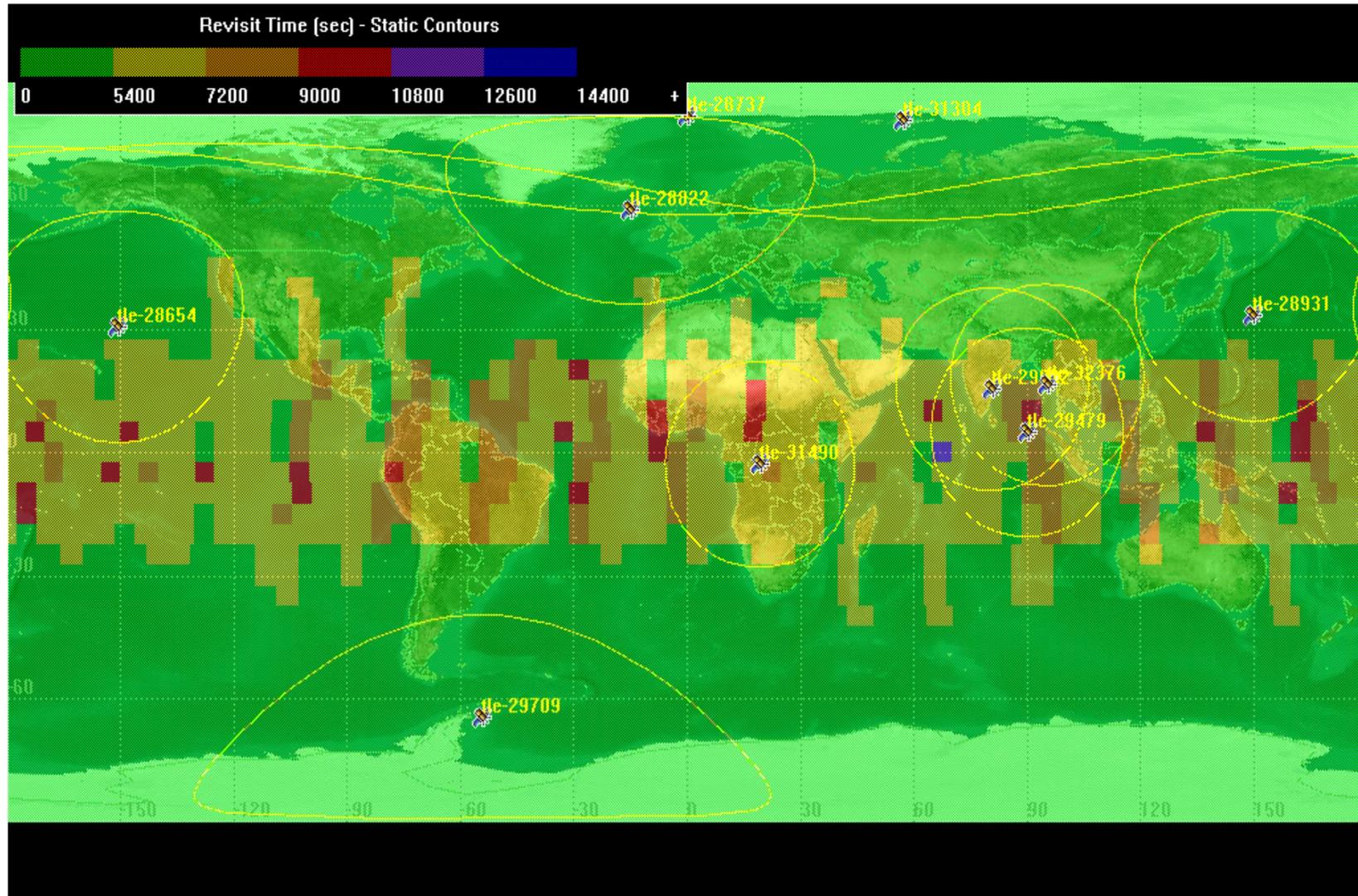
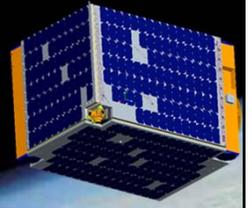
- Expert domain teams trace subsystem design variables to subsystem attributes using their own tools.
- Many attributes are multidimensional, e.g., latency is combination of:
  - Time from data collection to downlink
  - Time taken to downlink the data set
  - Time to taken distribute to data from ground station to the user
  - Time to process data
- Not always straightforward to define metrics. E.g., revisit time varies by:
  - Satellite
  - Orbital parameters
  - Orbital epoch
  - Location of interest on Earth
  - Transient factors (weather, duty cycle, etc.)
- Worst case, best case, mean, median, modal, etc., or combination could be considered most relevant performance metric.

# Utility Attributes



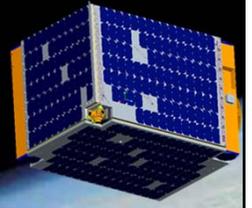
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# Example: Maximum Revisit Time



Attributes &  
design variables

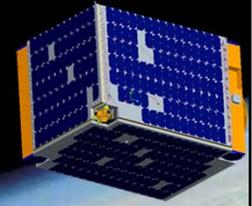
# Satellite Constellation Attributes



Utility attribute	Contributing subsystems
Refresh time globally	<ul style="list-style-type: none"> <li>• Orbit</li> <li>• Launch (only as an orbit driver)</li> <li>• Ground</li> </ul>
Refresh time over particular area of interest	<ul style="list-style-type: none"> <li>• Orbit</li> <li>• Launch (only as an orbit driver)</li> <li>• Ground</li> </ul>
Time on target	<ul style="list-style-type: none"> <li>• Orbit</li> <li>• Launch (only as an orbit driver)</li> <li>• Payload (footprint, duty cycle)</li> </ul>
Latency	<ul style="list-style-type: none"> <li>• Orbit</li> <li>• Launch (only as an orbit driver)</li> <li>• Payload (data volume and format, downlink rate)</li> <li>• Ground (time to downlink, distribute, &amp; process data)</li> </ul>
Redundancy	<ul style="list-style-type: none"> <li>• Bus (reliability/availability)</li> <li>• Orbit (refresh time with loss of spacecraft)</li> <li>• Ground (refresh time with loss of ground station)</li> </ul>

Attributes & design variables

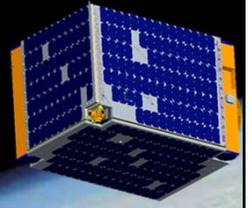
# Satellite Constellation Design Variables



System design variables	Subsystem design variables	Associated cost drivers
Spacecraft/bus	<ul style="list-style-type: none"> <li>• Number of spacecraft</li> <li>• Nanosat/microsat/smallsat/hosted payload</li> <li>• Controlled/uncontrolled</li> </ul>	<ul style="list-style-type: none"> <li>• Design &amp; build</li> <li>• Launch</li> <li>• Operations</li> </ul>
Orbital characteristics	<ul style="list-style-type: none"> <li>• Orbit planes</li> <li>• Eccentricities</li> <li>• Separation</li> <li>• Special cases (e.g., sun-synchronous)</li> </ul>	<ul style="list-style-type: none"> <li>• Launch</li> <li>• Propulsion &amp; delta-v</li> </ul>
Launch vehicle	<ul style="list-style-type: none"> <li>• Number of satellites per launch</li> <li>• Primary vs. secondary payload</li> <li>• Replacement availability</li> </ul>	<ul style="list-style-type: none"> <li>• Launch vehicles</li> <li>• Launch operations</li> </ul>
Payload	<ul style="list-style-type: none"> <li>• Selection of payload for mission</li> </ul>	<ul style="list-style-type: none"> <li>• Equipment design &amp; build cost</li> <li>• Operations</li> </ul>
Downlink	<ul style="list-style-type: none"> <li>• Frequency</li> <li>• Bandwidth</li> <li>• Geographic availability</li> </ul>	<ul style="list-style-type: none"> <li>• Equipment design &amp; build cost</li> <li>• Regulatory issues</li> </ul>
Redundancy, reliability, & replacement strategy	<ul style="list-style-type: none"> <li>• On-orbit (hot or cold) spares vs. replacement</li> <li>• Spacecraft reliability (expected lifespan)</li> </ul>	<ul style="list-style-type: none"> <li>• Initial cost vs. replacement cost</li> </ul>
Ground station	<ul style="list-style-type: none"> <li>• Number and placement of stations</li> <li>• Build vs. buy</li> <li>• Stationary vs. mobile</li> </ul>	<ul style="list-style-type: none"> <li>• Equipment design &amp; build cost</li> <li>• Start-up vs. operational costs</li> </ul>
Data processing & handling	<ul style="list-style-type: none"> <li>• Storage</li> <li>• Security</li> </ul>	<ul style="list-style-type: none"> <li>• Start-up costs</li> <li>• Operational costs</li> </ul>

Attributes & design variables

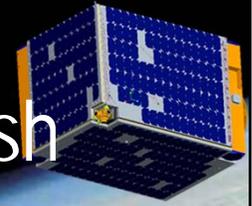
# Example Designs



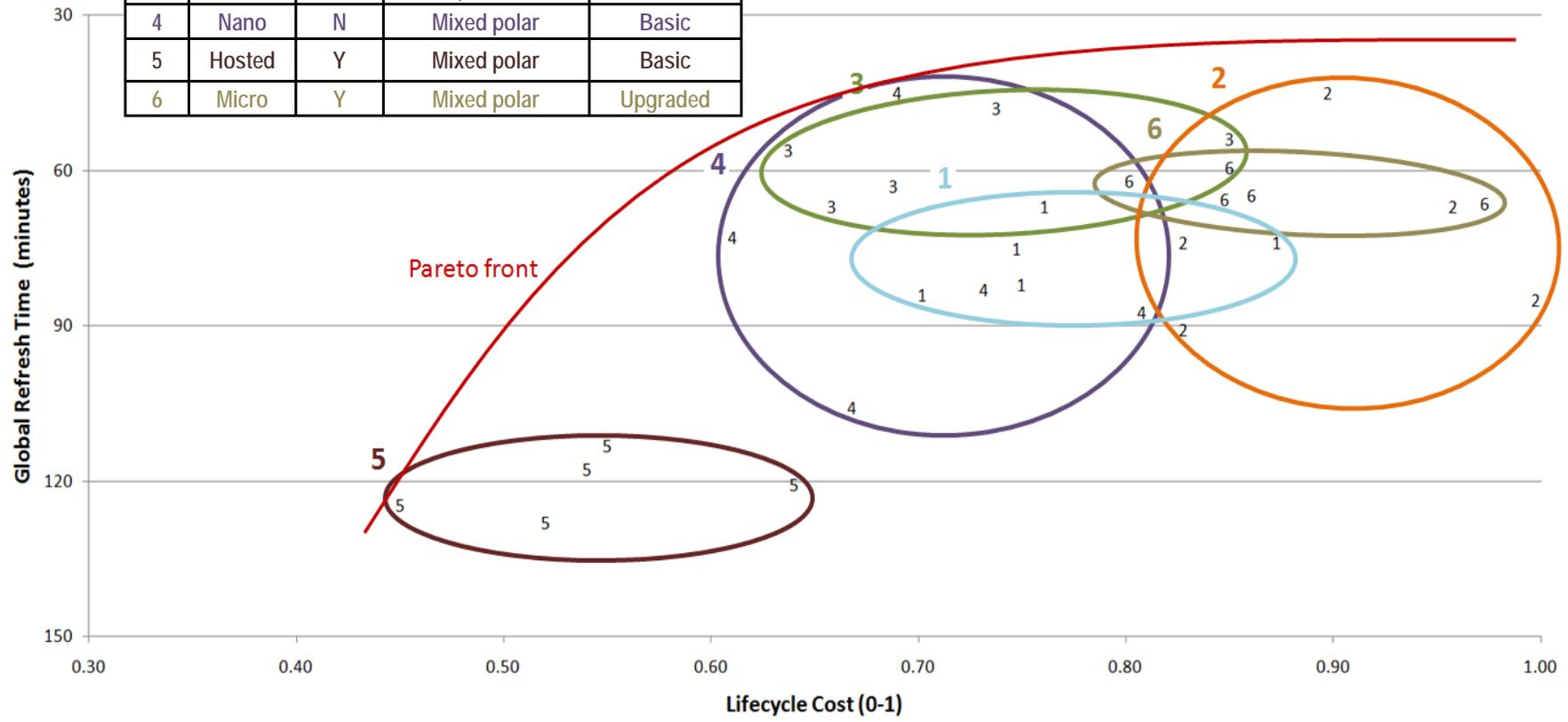
Design number	Bus	Propulsion?	Orbit type	Ground station configuration
1	Microsatellites	Y	Mixed polar	Basic
2	Microsatellites	N	Mixed polar	Basic
3	Microsatellites	N	Mixed including equatorial	Upgraded
4	Nanosatellites	N	Mixed polar	Basic
5	Hosted payload	Y	Mixed polar in historical locations	Basic
6	Microsatellites	Y	Mixed polar	Upgraded

Attributes & design variables

# Tradespace Exploration by Attribute: Global Refresh



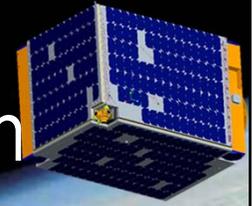
	Bus	Prop?	Orbit	Ground station
1	Micro	Y	Mixed polar	Basic
2	Micro	N	Mixed polar	Basic
3	Micro	N	Mixed including equatorial	Upgraded
4	Nano	N	Mixed polar	Basic
5	Hosted	Y	Mixed polar	Basic
6	Micro	Y	Mixed polar	Upgraded



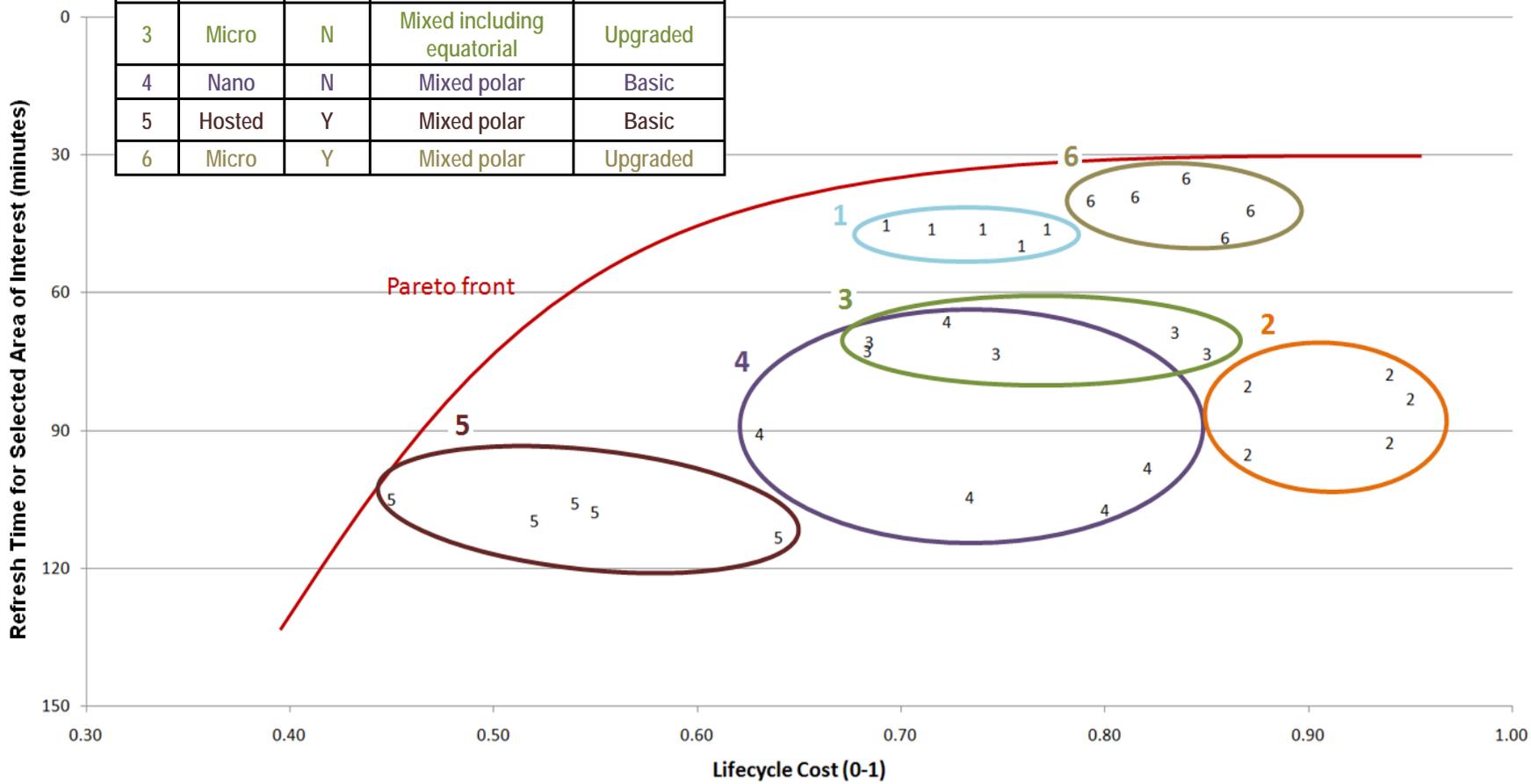
Tradespace Exploration



# Tradespace Exploration by Attribute: Local Refresh

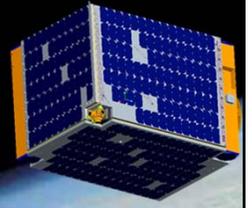


	Bus	Prop?	Orbit	Ground station
1	Micro	Y	Mixed polar	Basic
2	Micro	N	Mixed polar	Basic
3	Micro	N	Mixed including equatorial	Upgraded
4	Nano	N	Mixed polar	Basic
5	Hosted	Y	Mixed polar	Basic
6	Micro	Y	Mixed polar	Upgraded

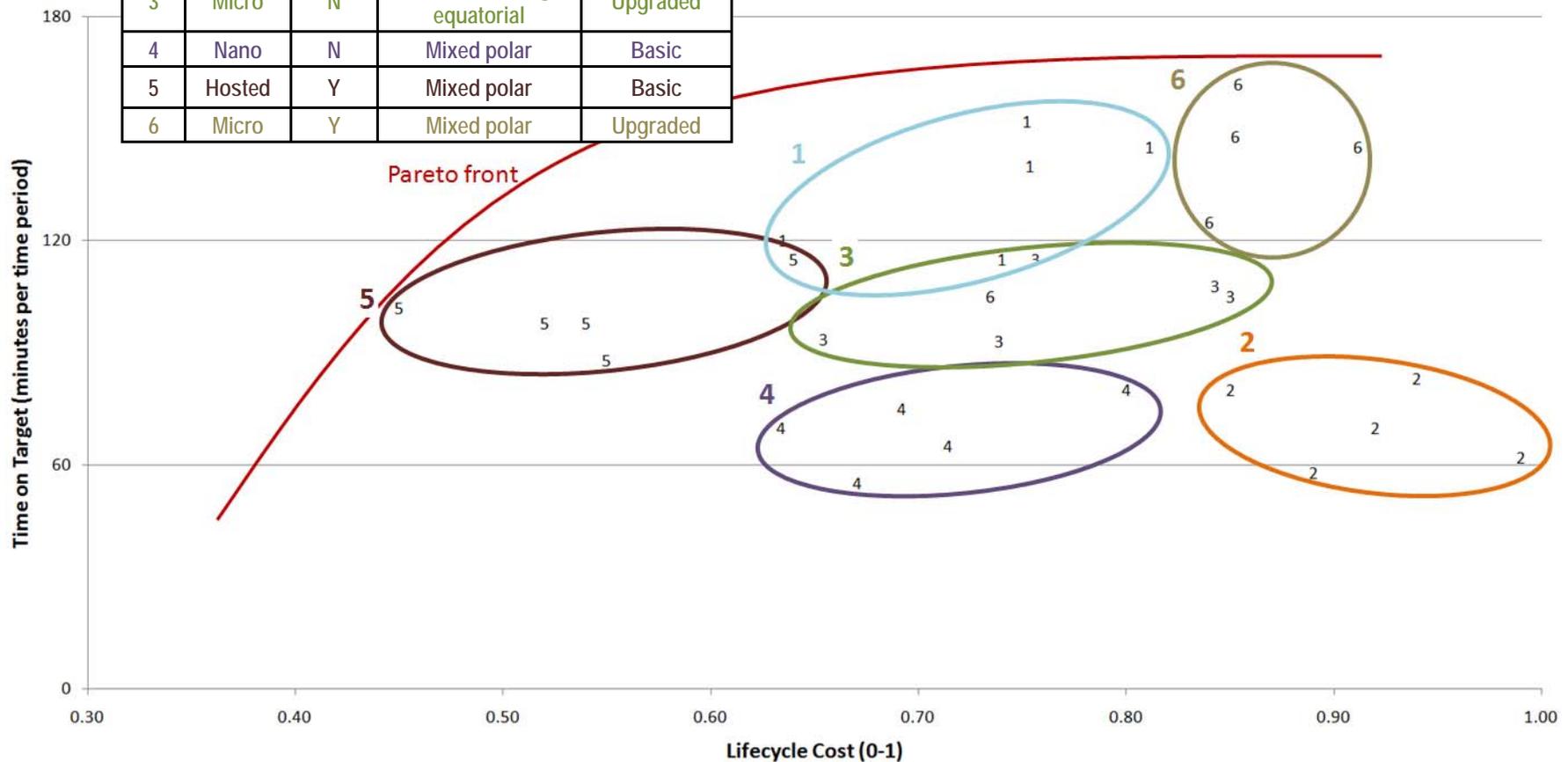


Tradespace Exploration

# Tradespace Exploration by Attribute: Observation Time

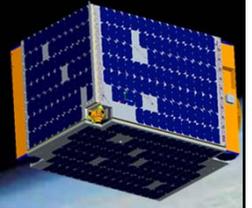


	Bus	Prop?	Orbit	Ground station
1	Micro	Y	Mixed polar	Basic
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6	Micro	Y	Mixed polar	Upgraded

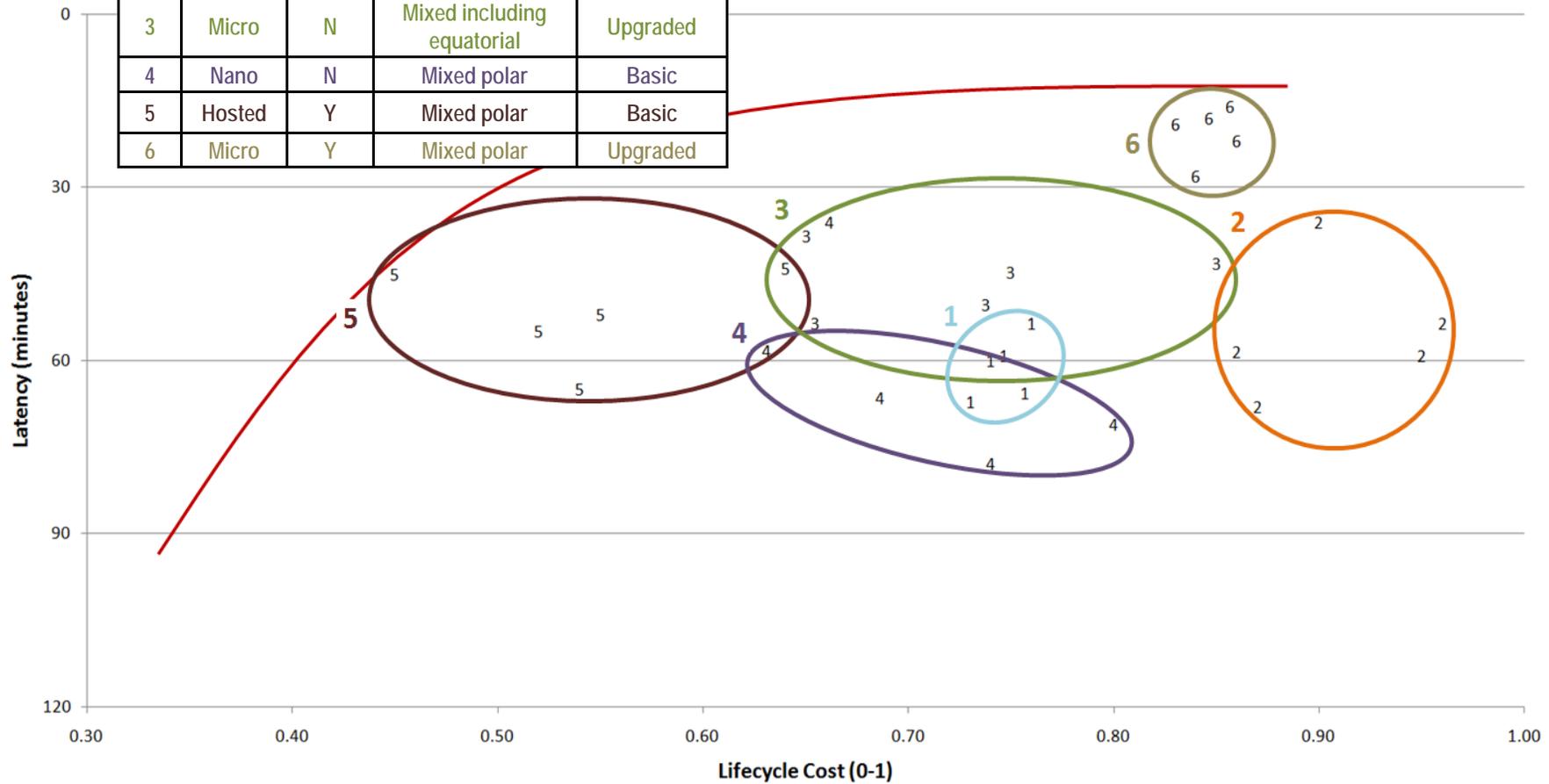


Tradespace Exploration

# Tradespace Exploration by Attribute: Observation Time

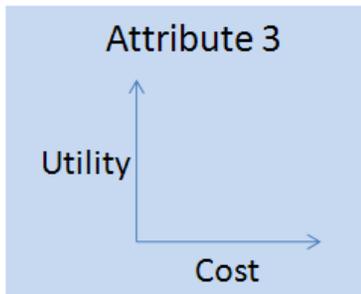
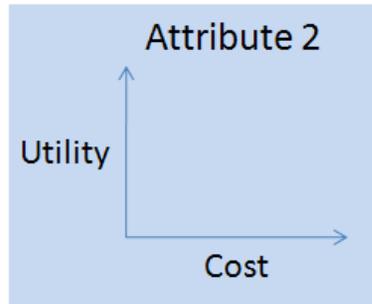
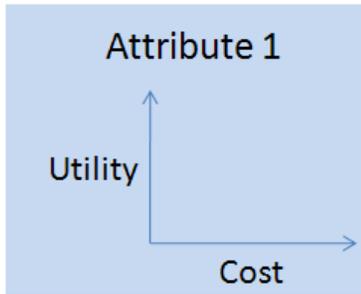
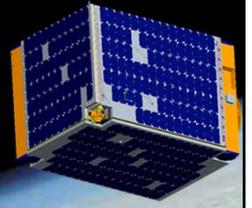


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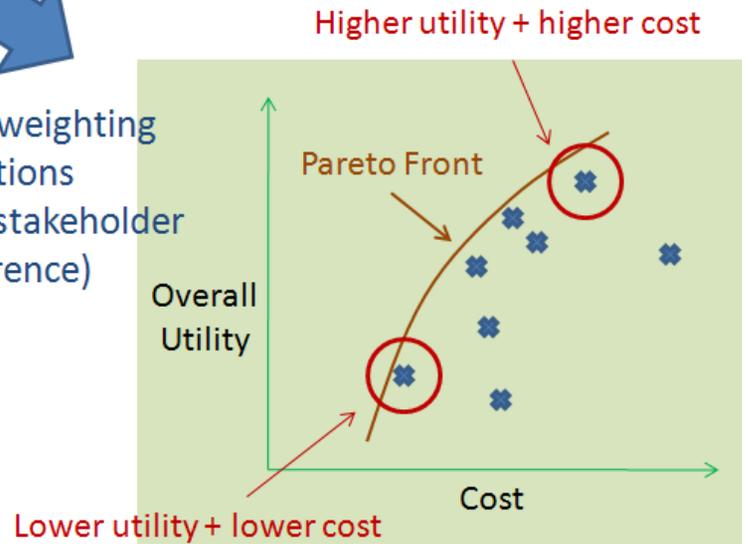


Tradespace Exploration

# MATE: Assembling the Multi-Attribute Tradespace

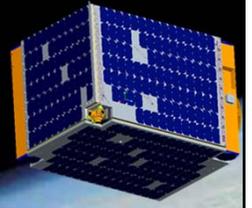


Attribute weighting functions  
(based on stakeholder preference)

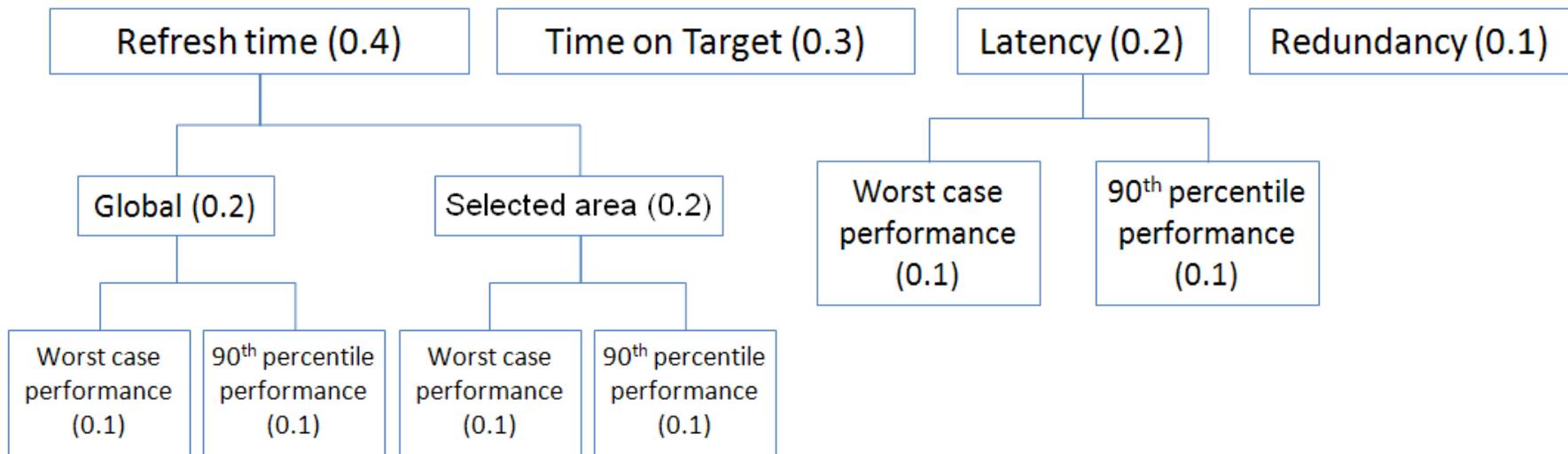


Tradespace  
Exploration

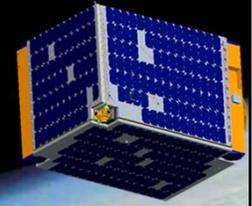
# MATE: Utility Weightings



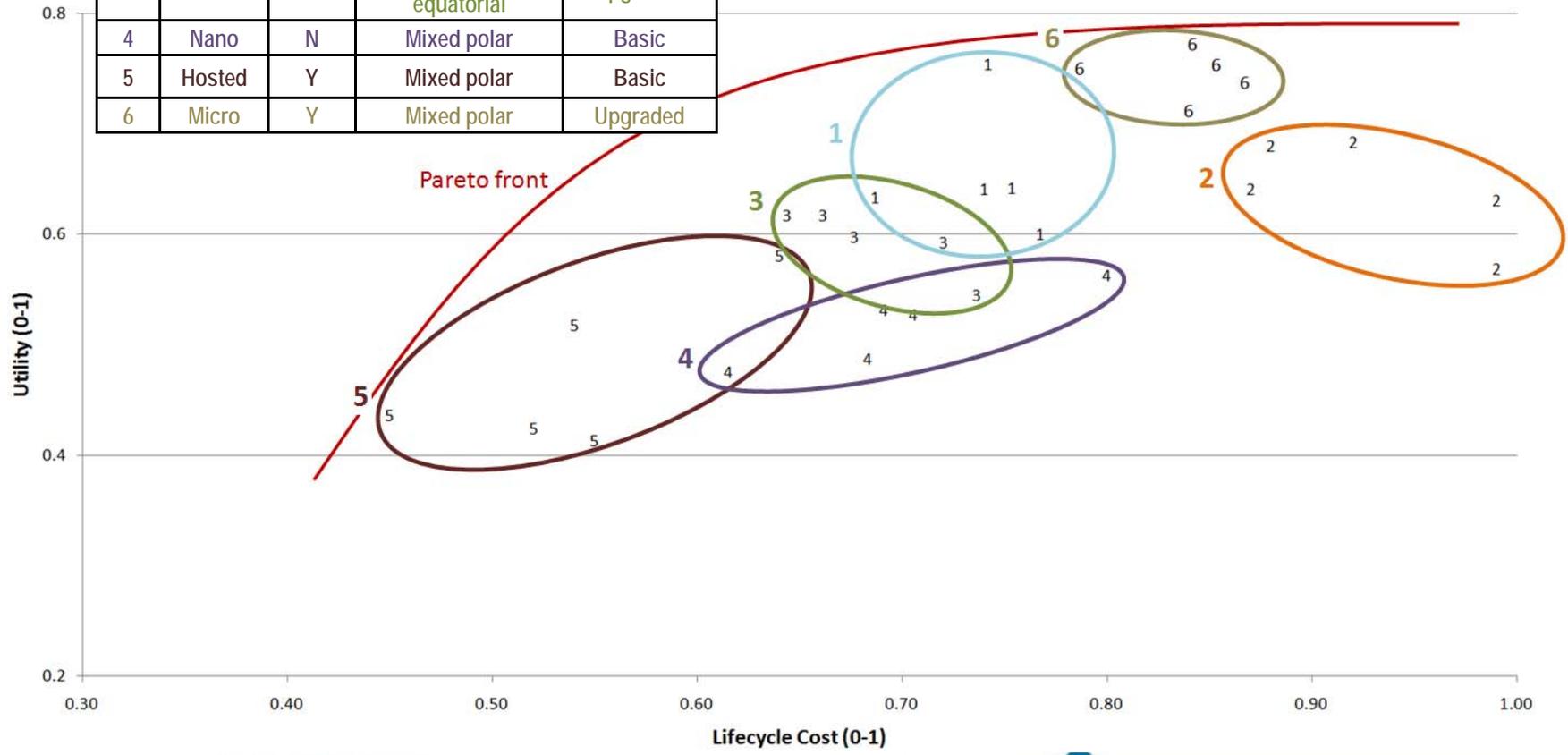
- Conjoint Analysis used to elicit utility weightings:
  - User selects preference from 2 “equal” alternatives of varying attributes multiple times.
  - Difference between nominal attribute rank and true attribute rank allows inference of attribute weightings.



# Multi-Attribute Tradespace

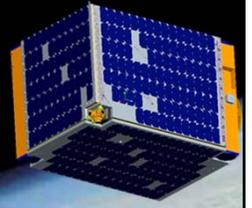


	Bus	Prop?	Orbit	Ground station
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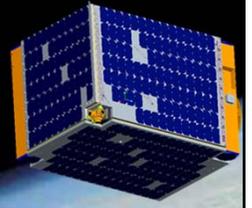
Tradespace Exploration

# Conclusions



- Tradespace is problem specific: process of assembling it can reveal predefined biases and lead to a re-examination of user requirements.
  - E.g., some designs that otherwise perform well, are excluded because they fail to meet a single requirement's minimum threshold.
- Engaging subsystem domain experts early:
  - Capitalizes on expert knowledge
  - Saves time
  - Reduces risk

# Conclusions



- Instead of generating a full system tradespace that might contain hundreds or thousands of potential designs, this method uses expert knowledge to generate a filtered subset containing only high value solutions
  - Performance optimizations at the subsystem level have already been performed in the assembled system level tradespace.
- Potential drawback: possible missed solutions in filtered tradespace.
- This bottom-up iterative approach is particularly appropriate for industry problems where it might be difficult to assemble a full computational system model.